

ABSTRACT OF THE DISCLOSURE

The invention relates to a hybrid blade (1) for thermal turbomachines, having an airfoil (2) made of a metallic material of a certain density, and having a blade root (3). It is characterized in that the blade root (3), compared with the airfoil (2), is made of a different metallic material having a lower density, and in that the airfoil (20) is connected to the blade root (3) in a positive-locking manner. The blade in this case is advantageously a compressor blade, in particular a high-pressure compressor blade, in which the airfoil (2) is made of a stainless CrNi steel and the blade root (3) is made of a high-temperature titanium alloy or an intermetallic gamma titanium aluminide alloy or an intermetallic orthorhombic titanium aluminide alloy.

(Fig. 1)